

FOR ESA'S COSMIC VISION PROGRAMME

http://jupiter-europa.cesr.fr/

Presentation to ESA's Solar System Working Group Oct. 11, 2007

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Many thanks to Astrium and CNES PASO for their support

| A science Road Map to the Jupiter system | | | | | | |
|--|--|--------------------------------|---|--------------|--|--|
| | Galileo | JUNO | LAPLACE | Europa | | |
| | 1st global exploration | 1st polar orbiter | 1st multi- platform mission | Lander | | |
| Jupiter | | Origin & int. Structure (1) | Int. Structure (2) | | | |
| | Science return limited | troposphere | Upper/middle atm. S and D | | | |
| Magneto -sphere | (data volume & Instruments) | Polar | Magnetodisk Satellite interactions | | | |
| Satellites | Key science questions identified | | Formation S & D (oceans?) System coupling | | | |
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Characterizing

ocean &

habitability

Search for

and life

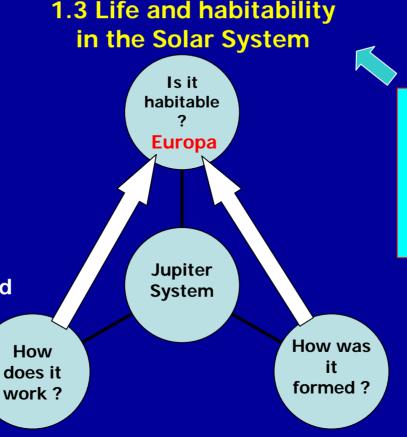
biosignatures

Europa

LAPLACE is tailored to address the themes of Cosmic Vision

2.1 From the Sun to the edge of the Solar System:

The Jovian magnetosphere as a local template for astrophysical magnetized disks and binary systems



CV theme 1:

What are the conditions for planet formation and the emergence of life?

1.1 From gas and dust to stars and planets.

2.2 Gaseous Giants and their Moons

2.3 The Building Blocks of the Solar System: Asteroids and Small Bodies

CV theme 2: How does the Solar System work?

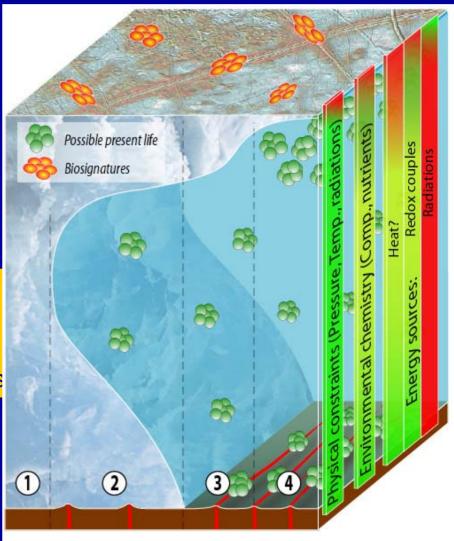
I- Is Europa habitable?

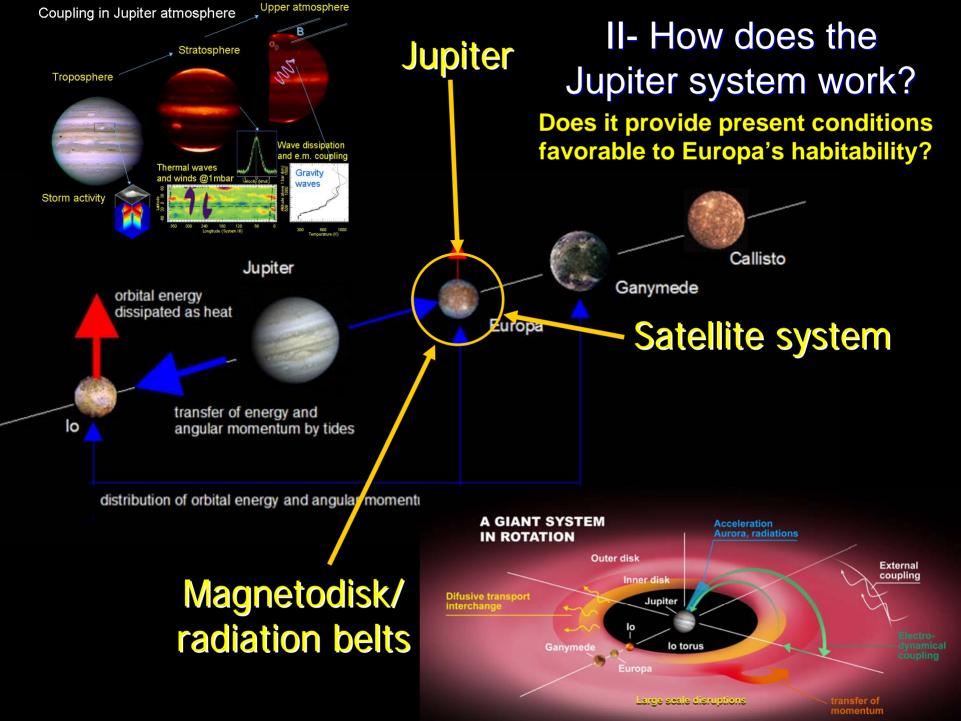
Highest Priority

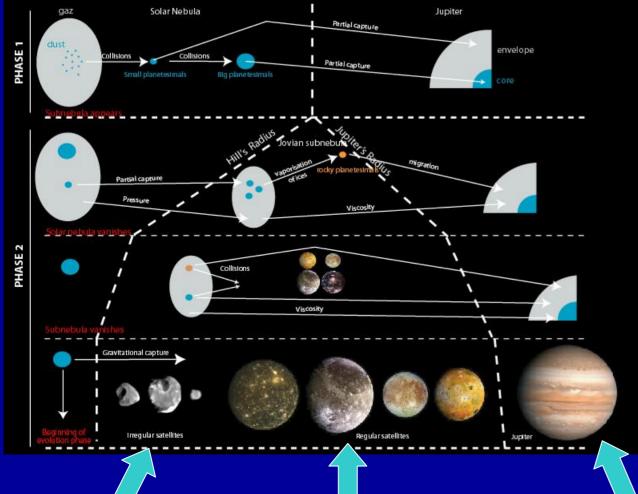
- Existence of a sub-surface ocean
- Surface composition and chemistry
- Characterization of the liquid layer
- Silicate/ocean surface topography
- Search for astrobiological signatures

Lower Priority

- Global surface morphology & dynamics
- Constraints on mantle dynamics
- Exosphere & Magnetospheric interactions







III- How did the Jovian system form ?

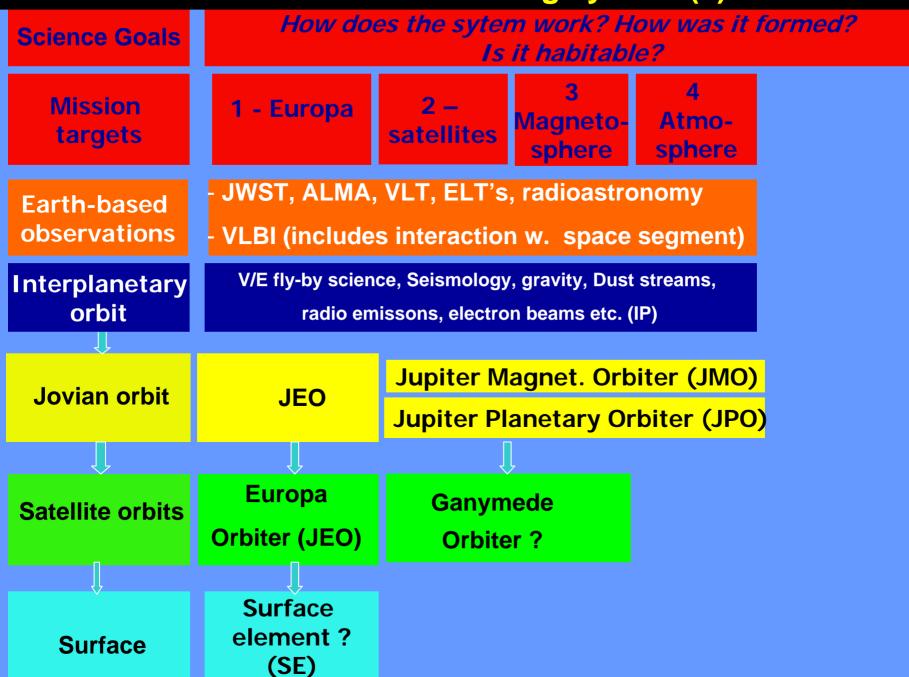
Did it provide

initial conditions favorable to Europa's habitability?

Constraints on early evolution of the system from the cratering records of irregular and regular satellites

Constraints on the protojovian nebula from chemical and isotopic abundances of the regular satellites Constraints on internal structure and core size (complementary to JUNO) using seismology

The LAPLACE observing system (1)



The LAPLACE observing system (2)

- two Jovian orbiters:

one 3-axis-stabilized platform for remote sensing instruments: JPO Jovian Planetary Orbiter

one spinning platform for in-situ fields-and particles measurements:

JMO Jupiter Magnetospheric Orbiter

one nadir-pointing platform:
 JEO Jupiter Europa orbiter

Two options for ESA L-class (Astrium support studies):

- Backup scenario : ESA-only mission.
 - One S/F launch.
 - Two s/c: Jupiter Relay Satellite (= JMO + JPO), JEO.
- Preferred scenario : Two medium capacity launchers with:
 - JPO + JMO on one launch JPO final injection to Ganymede orbit to be studied
 - JEO on the second launch
 - Many options for international collaboration

3 categories of scientific investigations required:

- measurements of the main planetary fields (Europa, other satellites)
- multi-spectral remote sensing of the surfaces, atmospheres and gas tori, and possibly sub-surfaces
- in situ remote sensing of the plasmas, fields, energetic particles gas and dust populations

Distributed between the platforms to:

- Maximize complementarities and synergies
- Take advantage of the specific pointing requirements of each platform

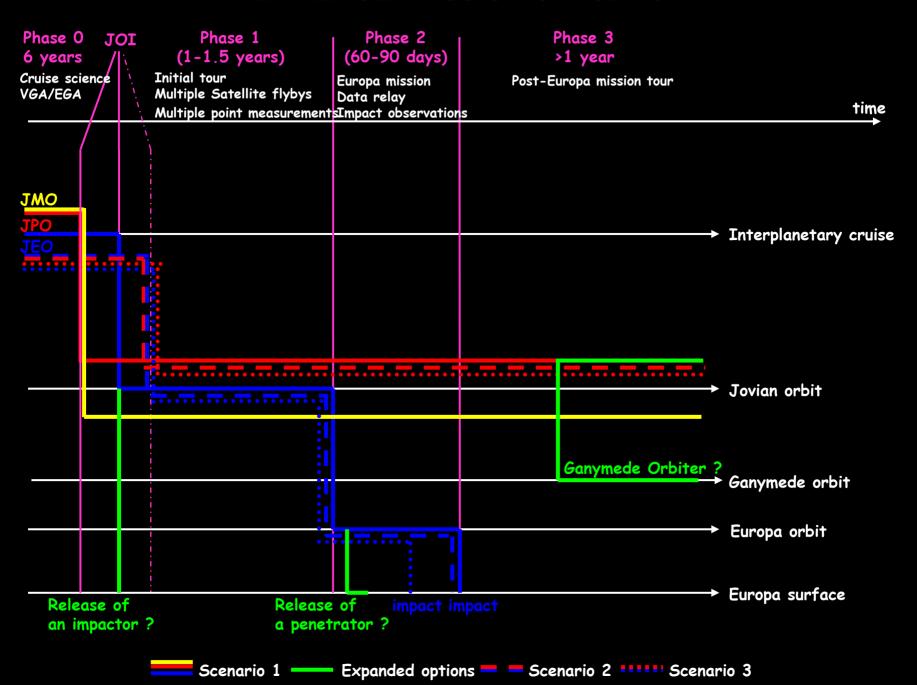
A possible science payload consists of:

- core instruments addressing high-priority science objectives,
- specific high-priority **instruments to be studied** during the assessment phase,
- complementary additional instruments.

A highly integrated payload approach will optimise science return

| | 1 | | <u> </u> | | | |
|------------------------------|--|---|--|--|--------------------------------------|--|
| | | Scenario 1 | | Scenario 1 | | |
| | | | | Integrated approach | | |
| | Payload mass (kg) 50/75 Shielding coefficient 0.6 | | | Payload mass (kg) 50/75 Shielding coefficient 0.6 | | |
| Jupiter Europa Orbiter | Core | Radio Science *Micro-Gradiometer Laser altimeter Magnetometers Camera package V/NIR spectrometer INMS Dust analyzer | 31538533 | Radio Science *Micro-gradiometer Laser altimeter/Camera int. Magnetometers Camera/NIR/UV integrated Integrated plasma package INMS Dust analyzer | 3 1 7 3 8 4 3 3 | |
| | TBS | Penetrating radar *Micro-gradiometer | 15 1 | Penetrating radar *Micro-gradiometer | 15 1 | |
| | Additional | X-ray imaging spectrometer Reduced plasma package ENA low-energy imager | | X-ray imaging spectrometer ENA low-energy imager | 5 2 | |
| | Addi | UV imaging spectrometer Y/neutron spectrometer Sub-mm wave sounder | 4 5 7 | y/neutron spectrometer Sub-mm wave sounder | 5 7 | |
| | | Payload mass (kg) 50, Shielding coefficient 0. | Payload mass (kg) 50/75 Shielding coefficient 0.8 | | | |
| Jupiter Planetary Orbiter | Core | Radio Science Magnetometers Camera package TIR imaging spectrometer V/NIR imaging spectrometer Sub-mm Wave Sounder Reduced plasma package | 3 2 15 4 5 7 | Radio Science Magnetometers Camera/NIR/UV integrated TIR imaging spectrometer Sub-mm Wave Sounder Integrated plasma package X-ray imaging spectrometer | 3 2 15 4 7 6 | |
| | TBS | Doppler spectro-imager | 5 | Doppler spectro-imager | 5 | |
| | Additional | ENA high-energy imager X-ray imaging spectrometer UV imaging spectrometer | 3 5 4 | ENA high-energy imager | 3 | |
| £ | ¥ | Dust analyzer | 3 | Dust analyzer | 3 | |
| rbiter | Payload mass (kg) 25/30 Shielding coefficient 0.8 | | | Payload mass (kg) 25/30 Shielding coefficient 0.8 | | |
| Jupiter Magnetospheric O | Core | Magnetometers Plasma spectrometers Energetic particle detectors Radio & Wave instrument | 2 7 6 5 | Magnetometers Integrated plasma package Radio & Wave instrument Dust Analyzer | 2 10 5 3 | |
| | Additional | Dust Analyzer | 3 | | | |

LAPLACE mission timeline



Key technology areas to be addressed in study phase

- Energy source: solar panels vs. RPS?
- Radiation tolerance of on-boars systems and instruments
- Planetary protection compliance (for JEO)
- Tight mass/energy budget, calls for specific integrated strategy
- Optimisation of data storage/transmission
- High navigation/positioning accuracy, a key element for Europa and the satellites (gravity, topography, ocean detection, internal structure, Laplace resonance)

To be assessed during first study phase, but no show-stoppers at preliminary study

Mission scenarios and international collaboration

- LAPLACE is feasable by ESA only (scenario 3), by narrowing-down to main science target: Europa characterisation + its coupling to the Jupiter system
- Broadly multi- and inter-disciplinary misison. Raises considerable interest for the exploration of Jupiter and Europa in the scientific communities of the U.S.A., Japan, Russia and beyond.
- Specific architecture (multi-platform, multi-target) provides an outstanding framework for international collaboration
- ESA has an ideal collaboration experience to lead the mission:
 - Cassini-Huygens with NASA
 - BepiColombo with JAXA
 - Long experience of collaboration with Russia in Europe
- ESA-JAXA-NASA collaboration, with RosCosmos participation, offers a unique potential to fly the mission in its most ambitious scenario:
 - Let us join our resources, our technology and science know-how,
 - And share science payloads on all three platforms via an international call for investigations
- We shall be ready for a 2018 launch!

Complements by Bob and Masaki on interests and contributions from U.S. and Japanese science communities

End of the presentation